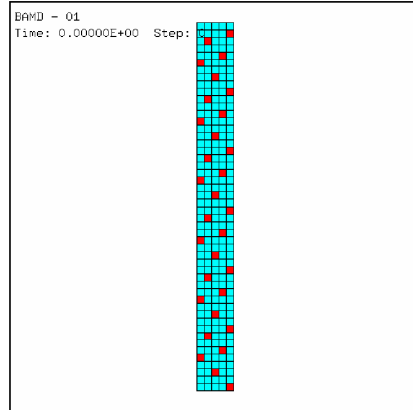


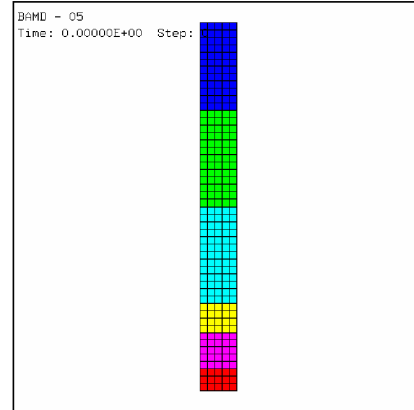
Example 15 – Bar Impact Revisited

- Taylor bar impact test:



Reference Case (no decomposition)

Case with time step partitioning (OPTI PART)



Case with domain decomposition (colors indicate sub-domains)
110

Problem description:

This example represents the bending of a thick beam in 3D conditions. The beam is clamped at the left end and loaded by concentrated forces at the right end. The material is linear elastic.

Two solutions are obtained. The first one without domain decomposition is used as a reference, while the second uses decomposition in three sub-domains, one of which is non-conforming with the other two.

Numerical Solutions

BAMD01

This calculation uses no domain decomposition and is used to obtain a reference solution. It uses a regular mesh of 306 elements of type Q41L and 219 elements of type Q42L.

The mesh generation file is:

```
*%siz 40
*
opti echo 1;
*
opti titr 'BAMD - 01';
opti dime 2 elem qua4;
*
p0=0 0;
p1=3.2E-3 0;
p2=0 32.4E-3;
tol=0.01E-3;
*
base=p0 d 5 p1;
stru=base TRAN 50 p2;
elim tol (stru et p2);
*
symax=stru poin droi p0 p2 tol;
top=stru poin droi p2 (p1 PLUS p2) tol;
*
str1=chan 'POI1' stru;
```

```
bas1=chan 'POI1' base;
viti=diff str1 bas1;
*
nstr=nbel stru;
nten=(nstr + 1) / 8;
nten=enti nten;
j=0;
*
i=0 - 3;repe loop nten;i=i+8;
si (<eg i nstr);
eli=stru elem i;
si (ega j 0);
sq42=eli;
sinon;
sq42=sq42 et eli;
finisj;
j=j+1;
finisj;
fin loop;
```

```
sq41=diff stru sq42;
*
mesh=stru et symax et viti;
tass mesh;
*
opti sauv form 'hamd01.msh';
sauv form mesh;
```

```
*
opti trac pac ftra 'hamd01_mesh.ps';
sq41=sq41 coul 'TURQ';
sq42=sq42 coul 'ROUG';
trac face (sq41 et sq42);
trac qual mesh;
fin;
```

The input file is:

```
BAMD - 01
$
ECHO
$VERI
$CONV win
CAST MESH
AXIS NONL
$
DIME
PTL 306 Q41L 219 Q42L 31 ZONE 2
TABL 1 5
  mtpo 7
  dep1 200
  tabl 1 2
TERM
$
GEOM Q41L SQ41 Q42L SQ42 TERM
$
COMP EPAI 1 TOUS
$
MATE VM23 RO 8930. YOUN 1.17D11 NU 0.35D0 ELAS 4.D8
  TRAC 2 4.D8 3.418803D-03 1.004D11 1000.003418803
  TOUS
$
CHAR 1 FACT 2
  DEPL 2 0.0 LECT BASE TERM
  DEPL 1 0.0 LECT SYMAX TERM
  TABL 2 0.0 1.0 1.0 1.0
$
INIT VITE 2 -227. LECT VITI TERM
$
ECRI DEPL VITE TPRE 10.E-6
```

```
POIN LECT P1 TOP TERM
fich alic tfre 80.e-8
$
OPTI NOTE
  csta 0.5
  log 1
$
CALC TINI 0. TEND 80.D-6
*****
SUIT
Post-treatment of bar impact problem
ECHO
*
RESU alic GARD PSCR
*
SORT GRAP
*
AXTE 1000.0 'Time [ms]'
*
COUR 1 'dr_p1' DEPL COMP 1 NOEU LECT p1 TERM
COUR 2 'dz_p2' DEPL COMP 2 NOEU LECT p2 TERM
COUR 3 'vr_p1' VITE COMP 1 NOEU LECT p1 TERM
COUR 4 'vz_p2' VITE COMP 2 NOEU LECT p2 TERM
*
TRAC 1 2 AXES 1.0 'DISPL. [M]' yzer
TRAC 3 4 AXES 1.0 'VELOC. [M/s]' yzer
*
QUAL DEPL COMP 1 LECT p1 TERM REFE 3.88425E-3 TOLE 1.E-3
  DEPL COMP 2 LECT p2 TERM REFE -1.09666E-2 TOLE 1.E-3
*****
FIN
```

BAMD03

This calculation uses the seame mesh as the previous one (no domain decomposition).
The time step partitioning technique is activated by specifying OPTI PART.

The mesh generation file is:

```
*%siz 40
*
opti echo 1;
*
opti titr 'BAMD - 03';
opti dime 2 elem quat;
*
p0=0 0;
p1=3.2E-3 0;
p2=0 32.4E-3;
tol=0.01E-3;
*
base=p0 d 5 p1;
stru=base TRAN 50 p2;
elim tol (stru et p2);
*
symax=stru poin droi p0 p2 tol;
top=stru poin droi p2 (p1 PLUS p2) tol;
*
str1=chan 'POI1' stru;
bas1=chan 'POI1' base;
viti=diff str1 bas1;
*
nstr=nbel stru;
nten=(nstr + 1) / 8;
nten=enti nten;
j=0;
```

```
*
i=0 - 3;repe loop nten;i=i+8;
si (<eg i nstr);
  eli=stru elem i;
  si (ega j 0);
  sq42=eli;
  sinon;
  sq42=sq42 et eli;
  fin1;
  j=j+1;
  fin1;
fin loop;
*
sq41=diff stru sq42;
*
mesh=stru et symax et viti;
tass mesh;
*
opti sauv form 'hamd03.msh';
sauv form mesh;
*
opti trac pac ftra 'hamd03_mesh.ps';
sq41=sq41 coul 'TURQ';
sq42=sq42 coul 'ROUG';
trac face (sq41 et sq42);
trac qual mesh;
fin;
```

The input file is:

```
BAMD - 03
$
ECHO
$VERI
$CONV win
CAST MESH
AXIS NONL
$
DIME
PTL 306 Q41L 219 Q42L 31 ZONE 2
TABL 1 5
  mtpo 7
  dep1 200
  tabl 1 2
TERM
$
GEOM Q41L SQ41 Q42L SQ42 TERM
$
COMP EPAI 1 TOUS
$
MATE VM23 RO 8930. YOUN 1.17D11 NU 0.35D0 ELAS 4.D8
  TRAC 2 4.D8 3.418803D-03 1.004D11 1000.003418803
  TOUS
$
CHAR 1 FACT 2
  DEPL 2 0.0 LECT BASE TERM
  DEPL 1 0.0 LECT SYMAX TERM
  TABL 2 0.0 1.0 1.0 1.0
$
INIT VITE 2 -227. LECT VITI TERM
$
ECRI DEPL VITE TPRE 10.E-6
```

```
POIN LECT P1 TOP TERM
fich alic tfre 80.e-8
$
OPTI PART NOTE step io
  csta 0.5
  log 1
$
CALC TINI 0. TEND 80.D-6
*****
SUIT
Post-treatment of bar impact problem
ECHO
*
RESU alic GARD PSCR
*
SORT GRAP
*
AXTE 1000.0 'Time [ms]'
*
COUR 1 'dr_p1' DEPL COMP 1 NOEU LECT p1 TERM
COUR 2 'dz_p2' DEPL COMP 2 NOEU LECT p2 TERM
COUR 3 'vr_p1' VITE COMP 1 NOEU LECT p1 TERM
COUR 4 'vz_p2' VITE COMP 2 NOEU LECT p2 TERM
*
TRAC 1 2 AXES 1.0 'DISPL. [M]' yzer
TRAC 3 4 AXES 1.0 'VELOC. [M/s]' yzer
*
QUAL DEPL COMP 1 LECT p1 TERM REFE 3.88425E-3 TOLE 1.E-3
  DEPL COMP 2 LECT p2 TERM REFE -1.09666E-2 TOLE 1.E-3
*****
FIN
```

BAMD05

This calculation uses the same mesh as the previous ones, but with decomposition into six sub-domains of various sizes. These are all conforming with respect to one another.

The scope is to activate a sort of sub-cycling in the sub-domains that undergo larger deformations than the others (i.e. the two sub-domains close to the projectile tip).

The mesh generation file is:

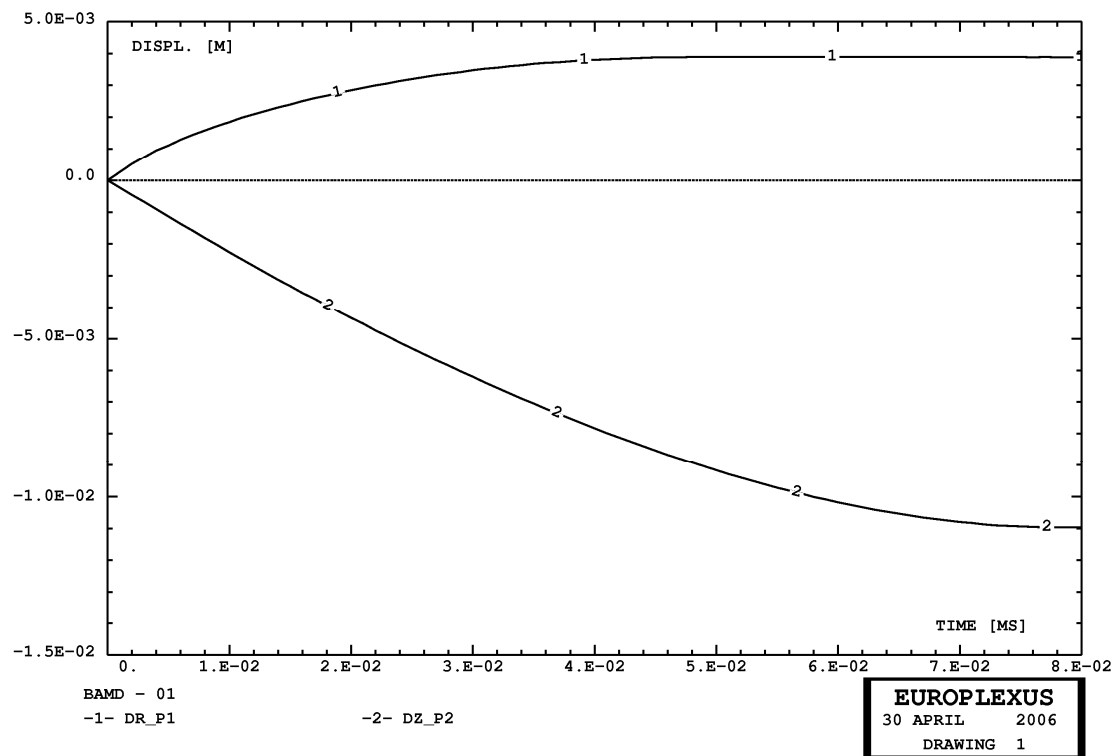
```
*%siz 40
*
opti echo 1;
*
opti titr 'BAMD - 05';
opti dime 2 elem qua4;
*
p0=0 0;
p1=3.2E-3 0;
p2=0 32.4E-3;
tol=0.01E-3;
*
base=p0 d 5 p1;
stru=base TRAN 50 p2;
elim tol (stru et p2);
*
symax=stru poin droi p0 p2 tol;
top=stru poin droi p2 (p1 PLUS p2) tol;
*
str1=chan 'PO11' stru;
bas1=chan 'PO11' base;
viti=diff str1 bas1;
*
nstr=nbcl stru;
nten=(nstr + 1) / 8;
nten=enti nten;
j=0;
*
i=0 - 3;repe loop nten;i=i+8;
  si (<eg i nstr);
  el=stru elem i;
  si (ega j 0);
  sq42=el1;
  sinon;
  sq42=sq42 et el1;
  fin1;
  j=j+1;
  fin1;
  fin loop;
  *
  sq41=diff stru sq42;
  *
  h1=2.1e-3;
  h2=5.1e-3;
  h3=8.1e-3;
  h4=16.2e-3;
  h5=24.3e-3;
  *
  n1=0;n2=0;n3=0;n4=0;n5=0;n6=0;
  i=0;repe loop2 (nbcl stru);i=i+1;
  el = stru elem i;
  bi = bary el;
  xb yb = coor bi;
  si (<eg yb h1);
  si (ega n1 0);
  si=ei;
  sinon;
  s1=s1 et ei;
  fin1;
  n1=n1+1;
  sinon;
  si (<eg yb h2);
  si (ega n2 0);
  *
  s2=ei;
  sinon;
  s2=s2 et ei;
  fin1;
  n2=n2+1;
  sinon;
  si (<eg yb h3);
  si (ega n3 0);
  s3=ei;
  sinon;
  s3=s3 et ei;
  fin1;
  n3=n3+1;
  sinon;
  si (<eg yb h4);
  si (ega n4 0);
  s4=ei;
  sinon;
  s4=s4 et ei;
  fin1;
  n4=n4+1;
  sinon;
  si (<eg yb h5);
  si (ega n5 0);
  s5=ei;
  sinon;
  s5=s5 et ei;
  fin1;
  n5=n5+1;
  sinon;
  si (ega n6 0);
  s6=ei;
  sinon;
  s6=s6 et ei;
  fin1;
  n6=n6+1;
  fin1;
  fin1;
  fin1;
  fin loop2;
  *
  list (nbcl s1);
  list (nbcl s2);
  list (nbcl s3);
  list (nbcl s4);
  list (nbcl s5);
  list (nbcl s6);
  list (nbcl stru);
  *
  mesh=stru et symax et viti;
  tass mesh;
  *
  opti sauv form 'bamd05.mesh';
  sauv form mesh;
  *
  opti trac pac ftra 'bamd05_mesh.ps';
  sq41=sq41 coul 'TURQ';
  sq42=sq42 coul 'ROU';
  trac face (sq41 et sq42);
  trac qual mesh;
  fin;
```

The input file is:

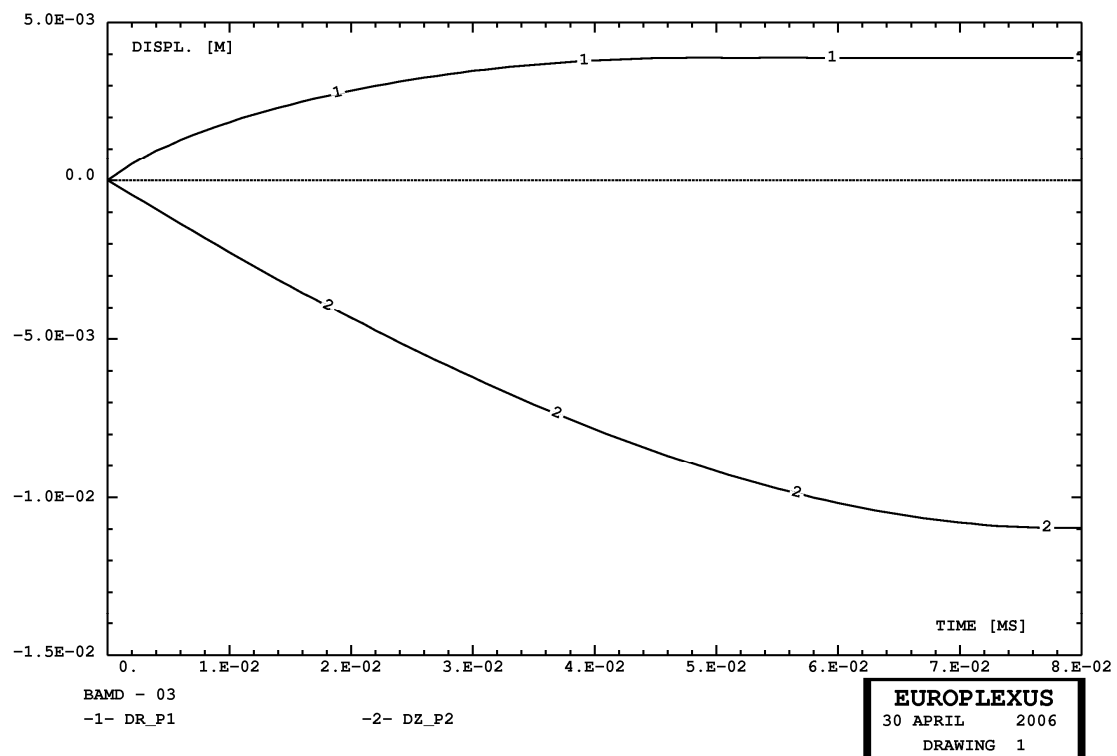
```
BAMD - 05
$
SCHO
SVERI
*CONV win
CAST MESH
AXIS NONL
$
DIME
  FTZL 306 Q41L 219 Q42L 31 ZONE 2
  mtpo 7
  TERM
  $
  GEOM Q41L SQ41 Q42L SQ42 TERM
  $
  COMP EPAI 1 TOUS
  $
  MATE VM23 RO 8930. YOUN 1.17D11 NU 0.35D0 ELAS 4.D8
  TRAC 2 4.D8 3.418803D-03 1.004D11 1000.003418803
  TOUS
  $
  LINK COUP
  BLOQ 2 LECT BASE TERM
  BLOQ 1 LECT SYMAX TERM
  $
  INIT VITE 2 -227. LECT VITI TERM
  $
  BCRI DEPL VITE TPRE 10.E-6
  POIN LECT P1 TOP TERM
  fich alic tpre 80.e-8
  $
  OPTI NOTE
  csta 0.5
  log 1

*
STRU 6
  DOMA LECT s1 TERM
  DOMA LECT s2 TERM
  DOMA LECT s3 TERM
  DOMA LECT s4 TERM
  DOMA LECT s5 TERM
  DOMA LECT s6 TERM
  $
  CALC TINI 0. TEND 80.D-6
  *=====
  SUIT
  Post-treatment of bar impact problem
  ECHO
  *
  RESU alic GARD PSCR
  *
  SORT GRAP
  *
  AXTE 1000.0 'Time [ms]'
  *
  COUR 1 'dr_p1' DEPL COMP 1 NOEU LECT p1 TERM
  COUR 2 'dz_p2' DEPL COMP 2 NOEU LECT p2 TERM
  COUR 3 'vr_p1' VITE COMP 1 NOEU LECT p1 TERM
  COUR 4 'vz_p2' VITE COMP 2 NOEU LECT p2 TERM
  *
  TRAC 1 2 AXES 1.0 'DISPL. [M]' yzer
  TRAC 3 4 AXES 1.0 'VELOC. [M/S]' yzer
  *
  QUAL DEPL COMP 1 LECT p1 TERM REFE 3.88425E-3 TOL 1.E-2
  DEPL COMP 2 LECT p2 TERM REFE -1.09666E-2 TOL 1.E-2
  *=====
  FIN
```

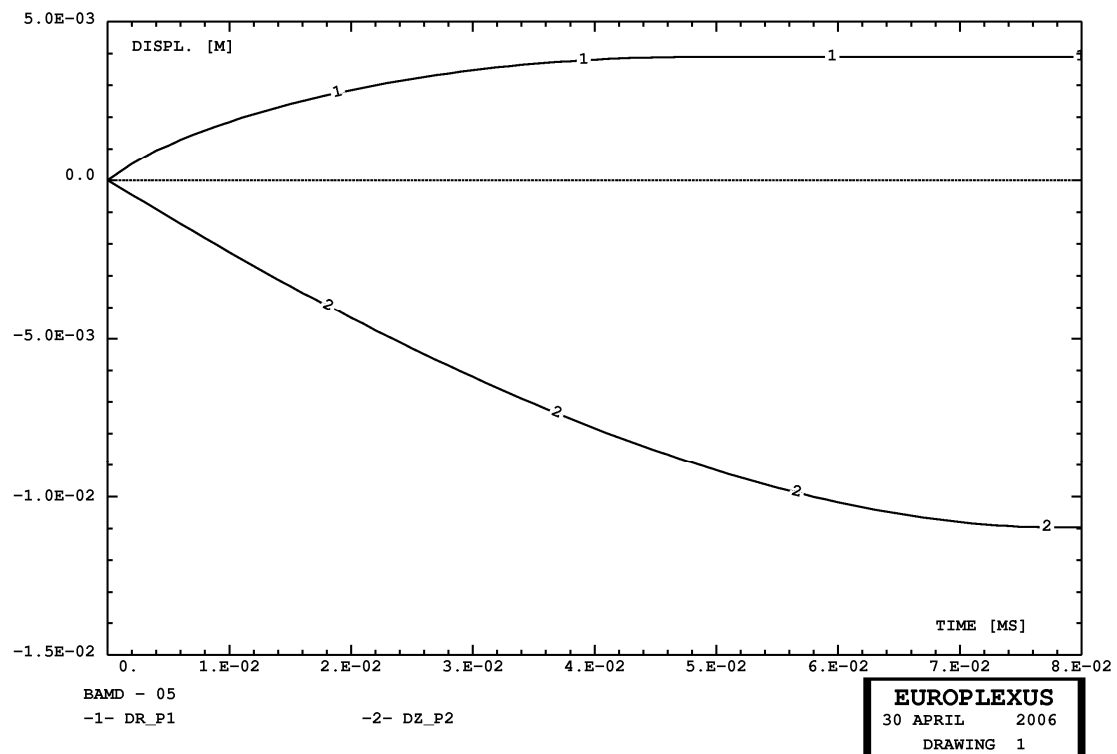
The characteristic displacements in the reference case are:



The displacements in the case with OPTI PART are:



The displacements in the case with domain decomposition are:



The final yield stress field in the reference case is:

BAMD - 01
Time: 8.00000E-05 Step: 61428

MAX: +6.64E+08

---: +6.60E+08

---: +6.40E+08

---: +6.20E+08

---: +6.00E+08

---: +5.80E+08

---: +5.60E+08

---: +5.40E+08

---: +5.20E+08

---: +5.00E+08

---: +4.80E+08

---: +4.60E+08

---: +4.40E+08

---: +4.20E+08

---: +4.00E+08

MIN: +4.00E+08

Ecro 7 [N/A]



The final yield stress field in the case with OPTI PART is:

BAMD - 03
Time: 8.00000E-05 Step: 1219

MAX: +6.64E+08
---: +6.60E+08
---: +6.40E+08
---: +6.20E+08
---: +6.00E+08
---: +5.80E+08
---: +5.60E+08
---: +5.40E+08
---: +5.20E+08
---: +5.00E+08
---: +4.80E+08
---: +4.60E+08
---: +4.40E+08
---: +4.20E+08
---: +4.00E+08
MIN: +4.00E+08
Ecro 7 [N/A]



The final yield stress field in the case with domain decomposition is:

BAMD - 05
Time: 8.00000E-05 Step: 1200

MAX: +6.65E+08
---: +6.60E+08
---: +6.40E+08
---: +6.20E+08
---: +6.00E+08
---: +5.80E+08
---: +5.60E+08
---: +5.40E+08
---: +5.20E+08
---: +5.00E+08
---: +4.80E+08
---: +4.60E+08
---: +4.40E+08
---: +4.20E+08
---: +4.00E+08
MIN: +4.00E+08
Ecro 7 [N/A]

